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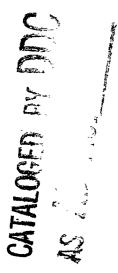
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Surface Impedance of a Thin Plasma Sheet

7 OCTOBER 1963

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Prepared for COMMANDER SPACE SYSTEMS DIVISION UNITED STATES AIR FORCE

Inglewood, California

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SURFACE IMPEDANCE OF A THIN PLASMA SHEET

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SURFACE IMPEDANCE OF A THIN PLASMA SHEET

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This technical documentary report has been reviewed and is approved for publication and dissemination. The conclusions and findings contained herein do not necessarily represent an official Air Force position.

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ABSTRACT

An analysis is made of electromagnetic propagation through a homogeneous, isotropic, and slightly ionized plasma slab. Exact solutions for the reflection and transmission coefficients are given as functions of slab thickness, incident angle, and plasma wave number. Two approximations are introduced which allow the plasma slab to become a thin current sheet. When the approximations are valid, the reflection and transmission coefficients for the thin plasma are a function of the sheet surface impedance only. This surface impedance is related simply to the product of the sheet thickness and conductivity.

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I. INTRODUCTION

Propagation of electromagnetic (EM) waves through plasma slabs is a topic of current interest. This situation is encountered in both plasma diagnostics and re-entry communications. A regime of particular interest in the re-entry application is that in which (1) the plasma wavenumber is large compared with the incident wavenumber, and (2) the slab thickness is small compared with the plasma skin depth. Such a thin plasma slab is shown to behave electrically as an inductive current sheet, having a lumped surface impedance consisting of an inductance in series with a resistance.

The EM properties of a thin plasma sheet can be represented completely by a surface impedance. That is to say, current sheets with the same surface impedance will produce the same EM reflection and transmission coefficients. This behavior of a thin plasma sheet enables it to be simulated easily. 1,2

A plane of equally spaced conducting wires and a thin conducting sheet with small apertures are two examples of current sheets with analogous surface impedances. 2-4

Also, arrays of fast switching microwave diodes behave as inductive current sheets over a large range of bias voltages. Switching diodes can be used to represent transient plasma sheets by varying the diode bias as a function of time.

Although this investigation has been motivated principally by plasma simulation interests, 5 this paper is concerned mainly with the EM mathematical model for a thin plasma sheet. The present analysis begins with

the derivation of the reflection and transmission coefficients for an ED wave obliquely incident upon a plasma slab of finite thickness. Two approximations of these coefficients are examined.

The first approximation assumes that the plasma wavenumber is large compared with the incident wavenumber. This restriction causes the refraction angle in the plasma to be near zero for all values of the incident angle and, in addition, simplifies the plasma wavenumber expression.

The second approximation involves the additional assumption that the slab thickness is small compared with the plasma skin depth. These requirements characterize a thin plasma sheet. A number of plots demonstrate the differences between the approximate and exact solutions.

Plasma surface impedance and conductivity are discussed in Section IV. A further examination of EM waves obliquely incident upon a thin plasma sheet is made from a different viewpoint. Starting with Maxwell's equations, the thin slab problem is resolved by applying the restrictions studied earlier as jump boundary conditions. In this approximation the reflection and transmission coefficients are seen clearly to be a function of only the sheet surface impedance.

All plasmas discussed in this paper are assumed to be homogeneous, isotropic and slightly ionized. Collision effects due to electron-neutral collisions are assumed to be velocity independent. No external magnetic fields are considered.

II. EXACT SOLUTION

In this section a plane EM wave obliquely incident upon a plasma slab is analyzed. When oblique incidence is encountered, it is advantageous to separate the radiation into perpendicular and parallel polarized components. Often the perpendicular polarization is called the transverse electric (TE) mode and has the electric field vector perpendicular to the plane of incidence. The plane of incidence is defined by the normal to the slab interface and the incident ray (the two sides which form angle θ in Fig. 1). Parallel polarization is also called the transverse magnetic (TM) mode and has the electric field vector parallel to the plane of incidence.

For the TE mode the appropriate plane wave solutions to Maxwell's equations for the electric field vectors in the three regions of Fig. 1 are

$$\overline{E}_{a} = \overline{a}_{x} \left\{ E_{1} \exp(-jk_{0}z \cos\theta) + E_{2} \exp(jk_{0}z \cos\theta) \right\} \exp[j(\omega t - k_{0}y \sin\theta)]$$
 (1)

$$\overline{E}_{b} = \overline{a}_{x} \left\{ E_{3} \exp(-jk_{p}z \cos\phi) + E_{4} \exp(jk_{p}z \cos\phi) \right\} \exp[j(\omega t - k_{p}y \sin\phi)]$$
 (2)

$$\overline{E}_{c} = \overline{a}_{x} \left\{ E_{5}(-jk_{o} \cos \theta) \right\} = \exp[j(\omega t - k_{o} y \sin \theta)]$$
(3)

where

$$k_0 = \omega(\mu_0 \varepsilon_0)$$
 (free space wavenumber) (4)

$$k_{\rm p} = k_{\rm o} \left(1 - \frac{j\sigma}{\omega \varepsilon}\right)^{1/2}$$
 (plasma wavenumber) (5)

$$\sigma = \frac{\omega_p^2 \varepsilon_0}{v_c^+ j\omega} \qquad \text{(plasma conductivity)} \tag{6}$$

 ω = incident frequency (rad/sec)

 ω_{p} = plasma frequency (rad/sec)

 $v_c = \text{collision frequency (rad/sec)}$

 ε_{o} = permittivity of free space

 μ_0 = permeability of free space

 \overline{a}_{x} = unit vector in x direction

 θ = angle of incidence

 ϕ = angle of refraction

t = time

x,y,z = coordinates

The letter subscripts refer to the total field in a given region, and the numeral subscripts designate in-going and out-going waves as shown in Fig. 1. All units are in the MKS system.

Two boundary conditions must be satisfied at each interface:

$$\left. \begin{array}{c}
\overline{E}_{a} = \overline{E}_{b} \\
\frac{\partial \overline{E}_{a}}{\partial z} = \frac{\partial \overline{E}_{b}}{\partial z}
\end{array} \right\} \text{ at } z = 0$$

$$\overline{E}_{b} = \overline{E}_{c}$$

$$\frac{\partial \overline{E}_{b}}{\partial z} = \frac{\partial \overline{E}_{c}}{\partial z}$$
at $z = d$

These conditions require the tangential components of the electric and magnetic fields to be continuous across each interface. In addition, Snell's law is needed to satisfy phase continuity at the boundaries and is stated as

$$\cos \phi = \left[1 - \left(\frac{k_o}{k_p}\right)^2 \sin^2\theta\right]^{1/2} \tag{7}$$

When the boundary conditions are applied to Eqs. (1)-(3), the following set of equations is obtained

$$\begin{pmatrix} E_1 \\ E_1 \\ 0 \\ 0 \end{pmatrix} = \begin{pmatrix} -1 & 1 & 1 & 0 \\ 1 & A & -A & 0 \\ 0 & 1/B & B & -C \\ 0 & -1/B & B & C/A \end{pmatrix} \begin{pmatrix} E_2 \\ E_3 \\ E_4 \\ E_5 \end{pmatrix}$$
(8)

where

$$A = \frac{k_p \cos \phi}{k_o \cos \theta} \tag{9}$$

$$B = \exp(jk_{p}d \cos\phi) \tag{10}$$

$$C = \exp(-jk_0 d \cos \theta)$$
 (11)

With the aid of Cramer's rule, the reflection and transmission coefficients are found from Eq. (8).

$$R(\perp) = \frac{E_2}{E_1}$$

$$R(1) = \frac{\frac{j}{2} \left(\frac{k_o \cos \theta}{k_p \cos \phi} - \frac{k_p \cos \phi}{k_o \cos \theta}\right) \sin \left[k_p d \cos \phi\right]}{\cos \left[k_p d \cos \phi\right] + \frac{j}{2} \left(\frac{k_o \cos \theta}{k_p \cos \phi} + \frac{k_p \cos \phi}{k_o \cos \theta}\right) \sin \left[k_p d \cos \phi\right]}$$
(12)

$$T(\perp) = \frac{E_5}{E_1}$$

$$T(1) = \frac{\exp(jk_0 d \cos \theta)}{\cos[k_p d \cos \phi] + \frac{j}{2} \left(\frac{k_0 \cos \theta}{k_p \cos \phi} + \frac{k_p \cos \phi}{k_0 \cos \theta}\right) \sin[k_p d \cos \phi]}$$
(13)

These are the coefficients for the perpendicular polarization (TE mode).

A similar procedure involving the magnetic field vectors leads to the coefficients for the TM mode. The proper solutions to Maxwell's equations for the magnetic field vectors in the three regions of Fig. 1 are

$$\overline{H}_{a} = \overline{a}_{x} \left\{ H_{1} \exp(-jk_{0}z \cos\theta) + H_{2} \exp(jk_{0}z \cos\theta) \right\} \exp[j(\omega t - k_{0}y \sin\theta)]$$
 (14)

$$\overline{H}_{b} = \overline{a}_{x} \left\{ H_{3} \exp(-jk_{p}z \cos\phi) + H_{4} \exp(jk_{p}z \cos\phi) \right\} \exp[j(\omega t - k_{p}y \sin\phi)] \quad (15)$$

$$\overline{H}_{c} = \overline{a}_{x} \left\{ H_{5} \exp(-jk_{0}z \cos\theta) \right\} \exp[j(\omega t - k_{0}y \sin\theta)]$$
(16)

The notation is the same as in the first case.

For the TM mode the boundary conditions which insure tangential continuity of the field across the interfaces are

$$\frac{\overline{H}_{a} = \overline{H}_{b}}{\frac{1}{j\omega\epsilon_{0}} \frac{\overline{H}_{a}}{\partial z} = \frac{1}{\sigma + j\omega\epsilon_{0}} \frac{\overline{H}_{b}}{\partial z}}$$
at $z = 0$ (17)

$$\frac{\overline{H}_{b} = \overline{H}_{c}}{\frac{1}{\sigma + j\omega\epsilon_{0}} \frac{\partial \overline{H}_{b}}{\partial z} = \frac{1}{j\omega\epsilon_{0}} \frac{\partial \overline{H}_{c}}{\partial z}}$$
at $z = d$ (18)

Snell's law has the same form as Eq. (7).

Applying the boundary conditions of Eqs. (17) and (18) to the plane wave solutions of Eqs. (14)-(16) leads to

$$\begin{pmatrix} H_1 \\ H_1 \\ 0 \\ 0 \end{pmatrix} = \begin{pmatrix} -1 & 1 & 1 & 0 \\ 1 & D & -D & 0 \\ 0 & 1/B & B & -C \\ 0 & -1/B & B & C/D \end{pmatrix} \begin{pmatrix} H_2 \\ H_3 \\ H_4 \\ H_5 \end{pmatrix}$$
(19)

where

$$D = \frac{k_0 \cos \phi}{k_p \cos \theta}$$

and B and C are given by Eqs. (10) and (11).

The reflection and transmission coefficients are obtained in the same manner as in the previous case

$$R(II) = \frac{E_2}{E_1} = \frac{-II_2}{H_1}$$
 (20)

$$R(||) = \frac{\frac{j}{2} \left(\frac{k_o \cos \phi}{k_p \cos \theta} - \frac{k_p \cos \theta}{k_o \cos \phi} \right) \sin \left[k_p d \cos \phi \right]}{\cos \left[k_p d \cos \phi \right] + \frac{j}{2} \left(\frac{k_o \cos \phi}{k_p \cos \theta} + \frac{k_p \cos \theta}{k_o \cos \phi} \right) \sin \left[k_p d \cos \phi \right]}$$

$$T(||) = \frac{E_5}{E_1} = \frac{H_5}{H_1}$$
 (21)

$$T(||) = \frac{\exp(jk_0 d \cos \theta)}{\cos\left[k_p d \cos \phi\right] + \frac{j}{2}\left(\frac{k_0 \cos \phi}{k_p \cos \theta} + \frac{k_p \cos \theta}{k_0 \cos \phi}\right) \sin\left[k_p d \cos \phi\right]}$$

These are the coefficients for the TM mode, and they differ from those for the TE mode only in the location of the quotient (cos θ /cos ϕ).

Equations (12), (13), (20), and (21) were programmed in the Fortran notation and evaluated on the 7090 computer. Figures 2-7 demonstrate the effect of the incident angle on the reflection and transmission coefficients. The coefficients are plotted as a function of slab thickness, $k_0 d$, for various values of θ , and fixed values of plasma ratios, ω_p/ω , and collision ratios, ν_c/ω .

For the lossless case $(v_c/\omega = 0)$, the zeros of the reflection coefficient are given by the expression

$$k_o d = \frac{n \pi}{\left[\cos^2 \theta - \left(\omega_p/\omega\right)^2\right]^{1/2}}, n = 1,2,3 \cdots$$

When the denominator of this expression is zero, a critical angle occurs. The oscillations in the coefficients cease when angle θ exceeds the critical value.

III. APPROXIMATE SOLUTIONS

Two sets of approximate solutions for Eqs. (12), (13), (20), and (21) are examined in this section. For the first approximation, it is assumed that the plasma wavenumber is much larger than the incident wavenumber.

$$\left|k_{p}\right|^{2} \gg k_{o}^{2} \tag{22}$$

This restriction allows Eq. (5) to become

$$k_p' = k_o \left[\frac{-j\omega_p^2}{\omega(v_c + j\omega)} \right]^{1/2}$$
 (23)

where k_p' is the asymptotic form of the plasma wavenumber. As $\left|k_p\right|^2$ becomes large, the refraction angle, ϕ , approaches a constant value of zero.

$$\cos \phi \approx 1$$
 (24)
$$\phi \approx 0$$

This condition follows from Eqs. (7) and (22).

The significance of this approximation lies in the simplified form of the wave solution in the plasma region as given in Eqs. (2) and (15). The elimination of the ϕ dependence represents an appreciable simplification when more complicated geometries are encountered.

If inequality (22) holds, the reflection and transmission coefficients become

$$R'(1) = \frac{\left(\frac{k_o \cos \theta}{o k_p^{\dagger}} - \frac{k_p^{\dagger}}{k_o \cos \theta}\right) \sin[k_p^{\dagger}d]}{\cos[k_p^{\dagger}d] + \frac{j}{2}\left(\frac{k_o \cos \theta}{o k_p^{\dagger}} + \frac{k_p^{\dagger}}{k_o \cos \theta}\right) \sin[k_p^{\dagger}d]}$$
(25)

$$T'(1) = \frac{\exp(jk_0 d \cos \theta)}{\cos[k_p' d] + \frac{j}{2} \left(\frac{k_0 \cos \theta}{k_p'} + \frac{k_p'}{k_0 \cos \theta}\right) \sin[k_p' d]}$$
(26)

$$R'(\parallel) = \frac{\left(\frac{k_o}{k_p^{\dagger} \cos \theta} - \frac{k_p^{\dagger} \cos \theta}{k_o}\right) \sin[k_p^{\dagger}d]}{\cos(k_p^{\dagger}d] + \frac{j}{2}\left(\frac{k_o}{k_p^{\dagger} \cos \theta} + \frac{k_p^{\dagger} \cos \theta}{k_o}\right) \sin[k_p^{\dagger}d]}$$
(27)

$$T'(||) = \frac{\exp(jk_0 d \cos \theta)}{\cos[k_p^{\dagger}d] + \frac{j}{2} \left(\frac{k_0}{k_p^{\dagger} \cos \theta} + \frac{k_p^{\dagger} \cos \theta}{k_0}\right) \sin[k_p^{\dagger}d]}$$
(28)

where primed quantities refer to the first approximation. The difference between approximate and exact reflection and transmission coefficients is shown in Figs. 8 and 9. These errors are measured by the

relationships

$$\Delta_{R}(\perp) = R'(\perp) - R(\perp)$$

$$\Delta_{T}(\perp) = T'(\perp) - T(\perp)$$

$$\Delta_{R}(||) = R'(||) - R(||)$$

$$\Delta_{T}(||) = T'(||) - T(||)$$
(29)

and are plotted as a function of align $k_0^{}$ d for various values of θ with plasma and collision ratios fixed. In Fig. (8) it is noted that the cos θ term reduces the effect of $|k_p^{}|^2$ being large for the TM mode and enhances this effect for the TE mode. As θ approaches 90°, the TM mode error increases and the TE mode error decreases.

Two assumptions are involved in the second approximation:^{8,9} (1) the plasma wavenumber is large compared with incident wave number,

$$|k_p|^2 \gg k_o^2 \tag{22}$$

(2) the slab thickness is small compared to the plasma skin depth such that the product

$$k_{p}^{d} \ll 1$$
 (30)

The first assumption was discussed in the preceding paragraphs. The second assumption allows the trigonometric functions to be replaced by the first term in their series expansions.

The second approximate solution represents the coefficients for a thin current sheet which is the principle topic under consideration here. For any plasma operating in cutoff $(k_p > k_o)$, the situation of primary interest is the one which can be approximated by a thin current sheet. Most of the other cases are not of interest, because the transmission coefficient is negligible.

If k d become much less than unity in Eqs. (25)-(28), the coefficients for the thin plasma sheet are found to be

$$R''(\perp) = \frac{-1}{1 + \frac{2k_0 \cos \theta}{jk_p^{'2} d}}$$
(31)

$$T''(\perp) = \frac{1}{1 + j \frac{k_p^{1/2} d}{2k_p \cos \theta}}$$
 (32)

$$R''(||) = \frac{-1}{2k_0}$$

$$1 + \frac{2k_0}{jk_0^{2}d \cos\theta}$$
(33)

$$T''(||) = \frac{1}{1 + j \frac{k_p^{1/2} \cos \theta}{2k_0}}$$
 (34)

Equations (33) and (34), the TM mode coefficients, are not valid as θ approaches 90 deg. This follows from the discussion concerning the angular error for the

first approximation. The correct limit for the TM mode coefficients can be obtained by analyzing Eqs. (20) and (21). From this analysis, it is found that the transmission coefficient approaches zero instead of one and that the reflection coefficient is greater than zero.

Figures (10a) and (10b) are plots of the magnitude and phase for the thin sheet coefficients. It should be noted that additional nomenclature has been introduced. From Eq. (23)

$$\frac{j k_p^{'2} d}{k_0} = \frac{w^2 D}{(V+j)}$$
 (35)

where

$$W = \frac{\omega_p}{\omega} \qquad \text{(plasma ratio)} \tag{36}$$

$$V = \frac{v_c}{\omega}$$
 (collision ratio) (37)

$$D = k_0 d \qquad (length ratio) \tag{38}$$

These dimensionless ratios are used for convenience. Angle θ does not appear directly in Figs. (10a) and (10b), but it can be introduced by modifying the length ratio to the forms

$$D_{\perp} = \frac{k_0 d}{\cos \theta} \tag{39}$$

$$D_{\parallel} = k_0 d \cos \theta$$

The four coefficients of Eqs. (31) - (34) are reduced to two since the TE and TM modes are specified by replacing D with D_{\perp} or $D_{||}$. Also, for ease in computation, the polar form of these equations is introduced so that

$$R'' = \frac{W^2D}{2} \sin \psi \exp[j(\pi - 4)]$$
 (40)

$$T'' = \frac{\sin \psi}{\sin \delta} \exp[j(\delta - 4)] \tag{41}$$

where

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$$tan \psi = \frac{1}{V + \frac{W^2D}{2}}$$
 (42)

$$tan \delta = \frac{1}{V}$$
 (43)

Errors in magnitude for the second approximation are seen by comparing Figs. (11) and (12) with (6) and (7). These figures demonstrate how error increases with increasing $k_0^{}$ and decreasing $\omega_p^{}/\omega$. The effect of incident angle θ is shown for both polarizations.

IV. SURFACE IMPEDANCE AND CONDUCTIVITY

Surface impedance is defined 7 as

$$Z_{s} = \frac{\overline{E}_{t}}{\overline{J}_{s}}$$
 (44)

where \overline{E}_t is the tangential electric field parallel to an incident surface of a slab and \overline{J}_s is the linear current density (A/m) flowing in the slab due to E_t . In terms of the current density, \overline{J} (A/m²), the surface current is

$$\overline{J}_{S} = \int_{0}^{d} (\overline{n} \times \overline{J}) \times d\overline{S}$$
 (45)

where \overline{n} is the unit normal to the incident surface, d is the slab thickness, and $d\overline{s}$ is the differential length in the direction of propagation. Ohm's law in terms of field quantities and conductivity is

$$\overline{E}_{t} = \frac{1}{\sigma} \overline{J} \tag{46}$$

If one substitutes Eq. (46) into (44)

$$Z_{s} = \frac{\overline{J}}{\sigma \overline{J}_{e}} \tag{47}$$

When the slab thickness is small compared to skin depth, the E field is constant through the slab in the direction of propagation. According to Ohm's law, the term \overline{J} also is constant in the same region. When it is integrated over the thickness d,

$$\overline{J}_{S} = \overline{J}d$$
 (48)

provided that ϕ is approximately zero for all θ . From Eqs. (47) and (48) the surface impedance of a thin sheet is

$$Z_{s} = \frac{1}{\sigma d} \quad (ohms/sq) \tag{49}$$

If the plasma is assumed to be slightly ionized and cold, its conductivity may be represented by the Lorentz gas model. ¹⁰ The governing equations for this model are

$$m \frac{d\overline{v}}{dt} + v_{c}\overline{v} = q\overline{E}$$

$$\overline{J} = \sigma\overline{E}$$

$$J = nq\overline{v}$$
(50)

where $\overline{\mathbf{v}}$ = electron velocity, m = electron mass, q = electron charge, and n = electron density.

If the E field is assumed to have a time dependence $exp(j\omega t)$, then the above equations lead to the expression

$$\sigma = \frac{nq^2}{m(v_c + j\omega)}$$
 (51)

for the plasma conductivity. Usually the plasma frequency $\boldsymbol{\omega}_p$ is introduced into Eq. (51) and

$$\sigma = \frac{\omega_p^2 \varepsilon_0}{v_c + j\omega} \tag{6}$$

where

$$\omega_{\rm p}^2 = \frac{n \, q^2}{m \, \epsilon_{\rm o}} \tag{52}$$

This is the expression stated at the beginning in Eq. (6).

If one replaces the conductivity in Eq. (49), the surface impedance becomes

$$Z_{s} = \frac{v_{c} + j\omega}{\omega_{p}^{2} \varepsilon_{o} d} = R_{s} + i X_{s}$$
 (53)

which is a complex impedance having a resistive component

$$R_{s} = \frac{v_{c}}{\omega_{p}^{2} \varepsilon_{o}^{d}}$$
 (54)

and a reactive component

$$X_{s} = \frac{1}{\omega_{p} \epsilon_{o} d}$$
 (55)

Equation (53) conveniently is expressed as

$$Z_{s} = Z_{o} \frac{V + j}{W^{2}D}$$
 (56)

by introducing the dimensionless parameters of Eqs. (36) - (38), and Z o is the intrinsic impedance of free space $(\mu_0/\epsilon_0)^{1/2}$. By simple substitution it follows that

$$R_{s} = \frac{Z_{o}V}{W^{2}D}$$
 (57)

$$X_s = \frac{Z_0}{w^2D}$$

The surface impedance of a thin sheet is given by Eq. (56) provided that assumptions (1) and (2) of the second approximation are valid. When one examines the components of the impedance, it is seen that the thin plasma behaves as an inductive current sheet.

V. JUMP PROBLEM

Now the thin sheet problem will be resolved by applying the restrictions of the second approximation directly to the plane wave solutions in regions a and c of Fig. 1. Imagine that the slab thickness d in Fig. 1 becomes small enough so that

$$k_{p}^{d} \ll 1$$
 (30)

If inequality (30) is satisfied, the tangential E fields in regions a and c can be related as

$$\overline{n} \times (\overline{E}_a - \overline{E}_c) = 0$$
 (58)

When the E field impinges upon the thin slab, it causes a surface current to flow. This surface current is equal to the discontinuity in the tangential H field.

$$\overline{J}_s = \overline{n} \times (\overline{H}_a - \overline{H}_c) = \frac{1}{\overline{Z}_s} = \overline{E}_t$$
 (59)

The surface current may be replaced by the surface impedance and the E_{t} field in inequalities (22) and (30) are valid. Inequality (22) insures that ϕ is near zero.

It will be necessary to find H given E and E given H for the two polarizations.

Maxwell's two curl equations furnish the required relations:

$$\overline{H} = \frac{j}{\omega \mu_{o}} \quad \overline{\nabla} \times \overline{E}$$

$$\overline{E} = \frac{1}{\sigma + j\omega \varepsilon_{o}} \overline{\nabla} \times \overline{H}$$
(60)

First, the TE mode is considered. At z=0 the jump conditions of Eqs. (58) and (59) are applied to the wave solutions of Eqs. (1) and (3), yielding

$$E_1 + E_2 = E_5$$
 (61)

$$\frac{\cos \theta}{Z_0} \left[E_1 - E_2 - E_5 \right] = \frac{1}{Z_5} E_5$$
 (62)

where Z_0 is the intrinsic impedance of free space. After algebraic manipulations of Eqs. (61) and (62), the reflection and transmission coefficients are found to be

$$R''(\perp) = \frac{-1}{2 Z_{s} \cos \theta}$$

$$1 + \frac{Z_{s} \cos \theta}{Z_{0}}$$
(63)

$$T''(1) = \frac{1}{\frac{Z_{o}}{2 Z_{c} \cos \theta}}$$

$$(64)$$

If Eqs. (31), (32), and (35) are compared with Eqs. (56), (63), and (64), the coefficients for the jump conditions are found to be the same as those for the second approximation.

In the case of TM mode the same procedure is followed. At z = 0 the jump conditions are applied to Eqs. (14) and (16) to obtain

$$- H_1 + H_2 = H_5$$
 (65)

$$H_1 + H_2 - H_5 = \frac{Z_0 \cos \theta}{2 Z_s} H_5$$
 (66)

These two equations can be solved for the TM mode coefficients which are

$$R''(\perp) = \frac{-1}{2 \frac{Z}{Z}}$$

$$1 + \frac{S}{Z \cos \theta}$$
(67)

$$T''(\perp) = \frac{1}{\frac{Z_0 \cos \theta}{2 Z_S}}$$
 (68)

When one compares Eqs. (33), (34), and (35) with Eqs. (56), (67), and (68), it is found that the jump solution and the second approximate solution are the same for the TM mode also.

Thus, it has been shown that the reflection and transmission coefficients of the plasma sheet are a function of the sheet surface impedance only. This fact enables a straightforward simulation of the thin plasma by the methods mentioned in the introduction.

VI. CONCLUSION

The first topic was an analysis of the exact solutions for the reflection and transmission coefficients describing propagation through plasma slab of finite thickness. Figures 2-7 are plots of the exact solution for these coefficients indicating their dependence on the plasma ratio, collision ratio, length ratio, and incident angle. An expression for determining the zeros of the reflection coefficients for the lossless case is given. From the denominator of this expression it is possible to find the critical angle of a lossless plasma.

Two approximations of the exact solution are examined. The first approximation was based on the assumption that the plasma wave number is large compared to the incident wavenumber. This assumption made the problem independent of the refraction angle. Plots were given which showed the difference between the first approximation and the exact solution. As the angle of incidence approaches 90 deg the TM mode error increased and the TE mode error decreased. Error in both modes decreased for larger plasma ratios.

The second approximation, which characterizes the jump conditions, involves the added assumption that the sheet thickness is small compared to the plasma skin depth. Plots for the magnitude and phase of the jump coefficients are given, but these curves do not indicate directly the θ dependence. Additional plots, which show the θ -dependence for the second approximation, may be compared directly with the plots for the exact solution. Once again an increasing error is found for the TM mode as θ approaches 90 deg. Error was found to become larger as $k_0 d$ increased and ω_p/ω decreased.

In the last section the problem was resolved using jump conditions. The results of the second approximation and the jump solution were compared and found to be identical. Therefore, it has been verified that a thin plasma sheet satisfying the jump conditions can be completely represented electrically by its surface impedance.

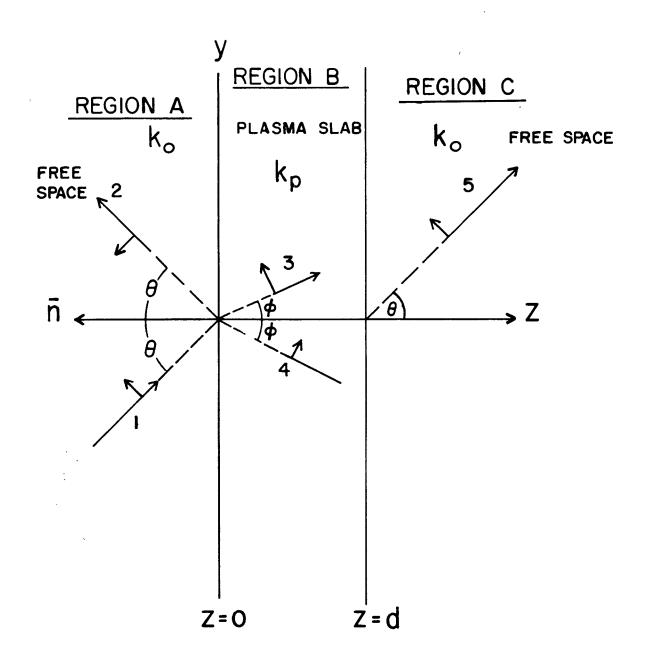


Fig. 1. Finite Plasma Slab

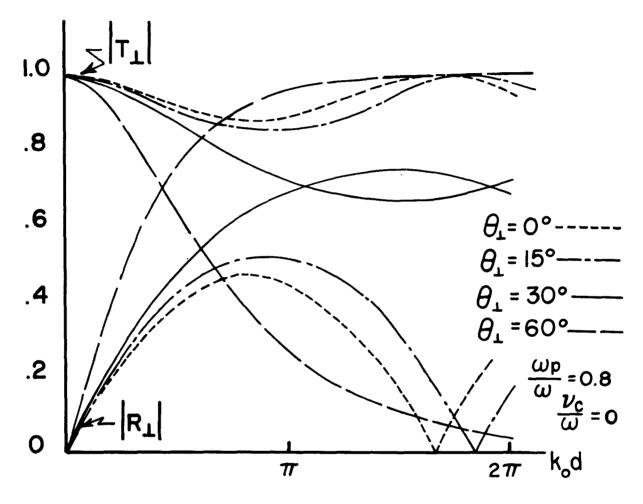


Fig. 2a. Exact Solution, above Cutoff, Lossless (TE Mode)

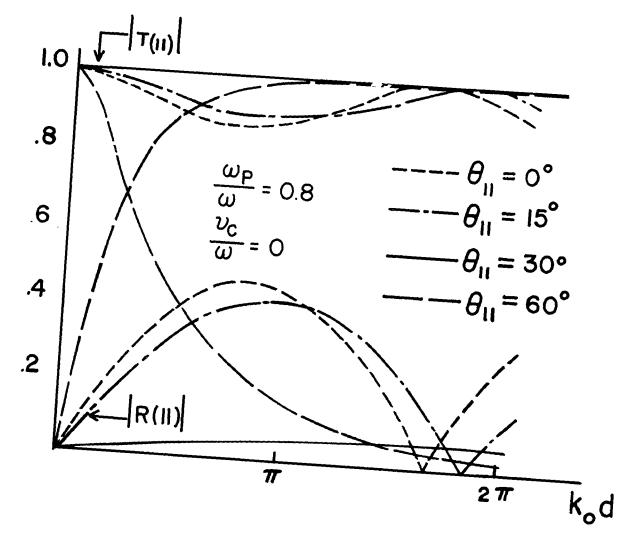


Fig. 2b. Exact Solution, above Cutoff, Lossless (TM Mode)

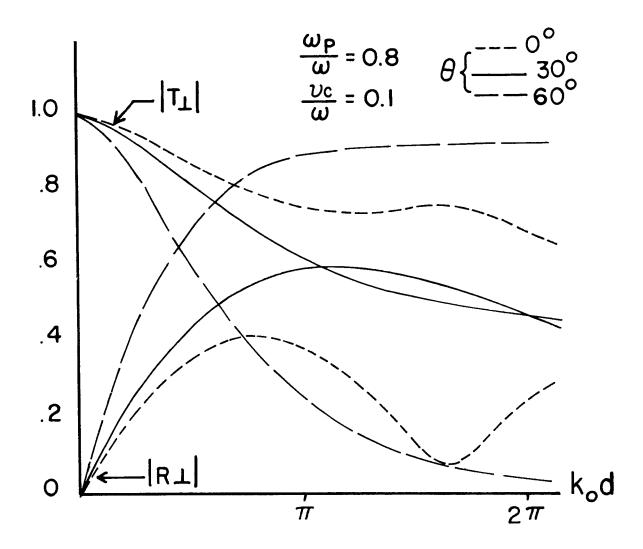


Fig. Sa. Exact Solution, above Cutoff, Lossy (TE Mode)

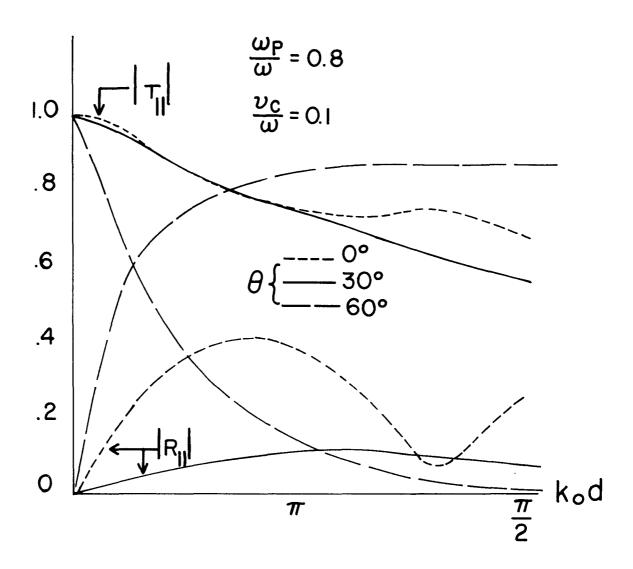


Fig. 3b, Exact Solution, above Cutoff, Lossy (TM Mode)

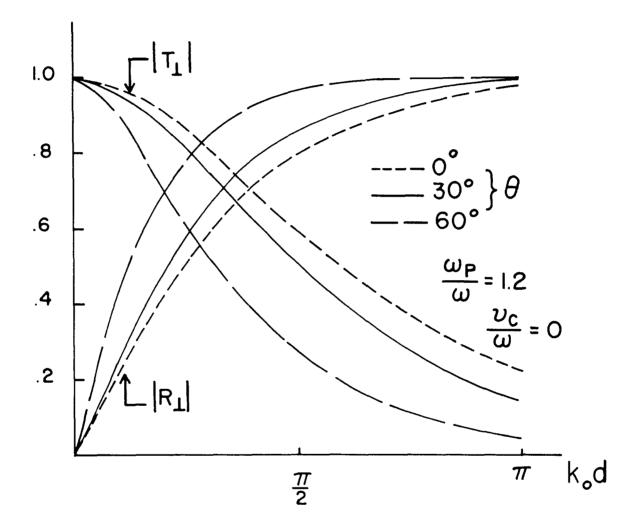


Fig. 4a. Exact Solution, below Cutoff, Lossless (TE Mode)

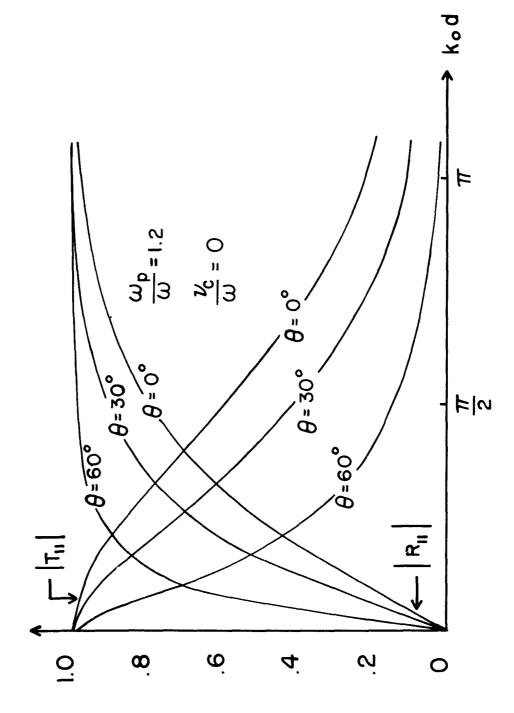


Fig. 4b. Exact Solution, below Cutoff, Lossless (TM Mode)

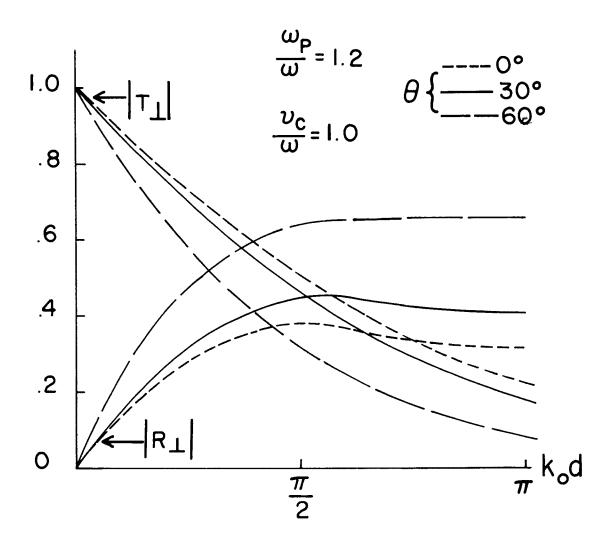


Fig. 5a. Exact Solution, below Cutoff, Lossy (TE Mode)

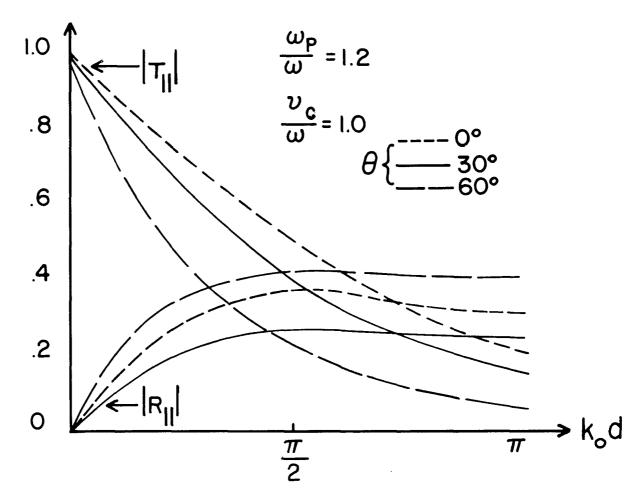


Fig. 5b. Exact Solution, below Cutoff, Lossy (TM Mode)

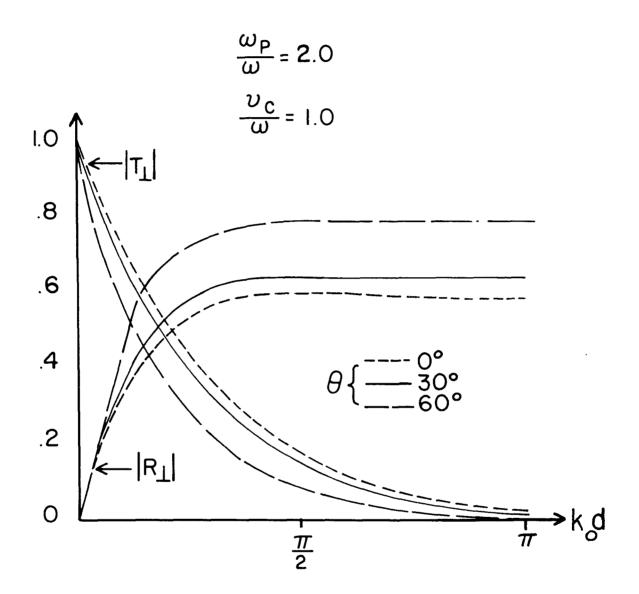


Fig. 6a. Exact Solution, below Cutoff, Lossy, ω_p/ω = 2 (TE Mode)

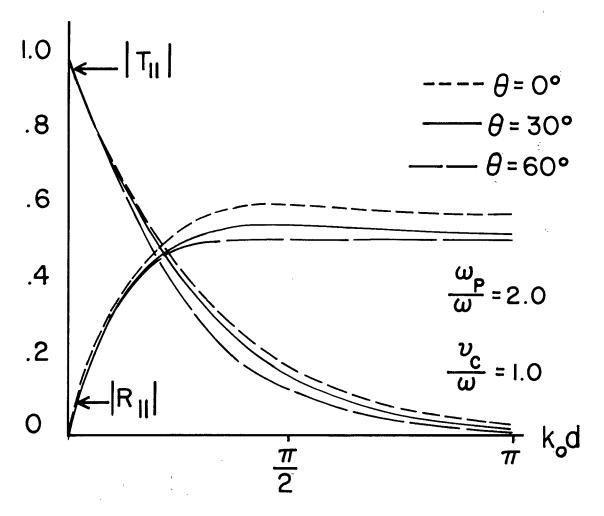


Fig. 6b. Exact Solution, below Cutoff, Lossy, $\omega_p/\omega = 2$ (TM Mode)

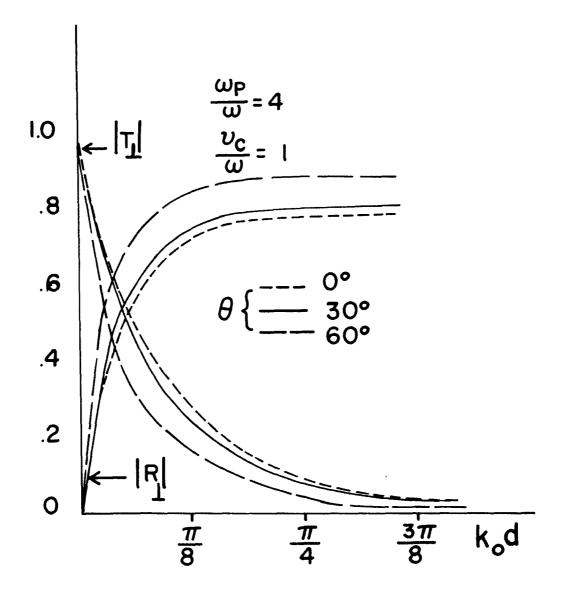


Fig. 7a. Exact Solution, below Cutoff, Lossy, $\omega_p/\omega = 4$ (TM Mode)

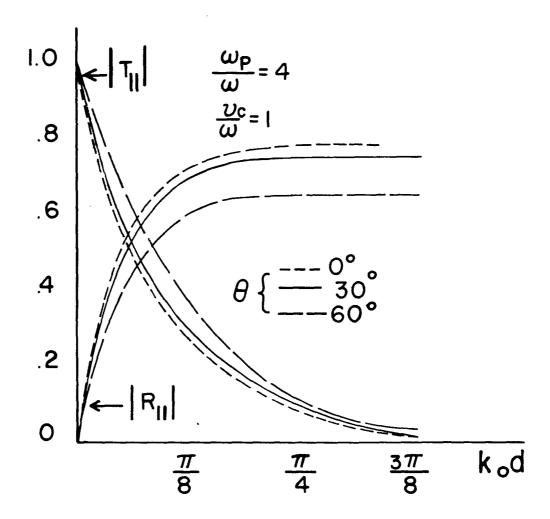


Fig. 7b. Exact Solution, below Cutoff, Lossy, $\omega_p/\omega = 4$ (TM Mode)

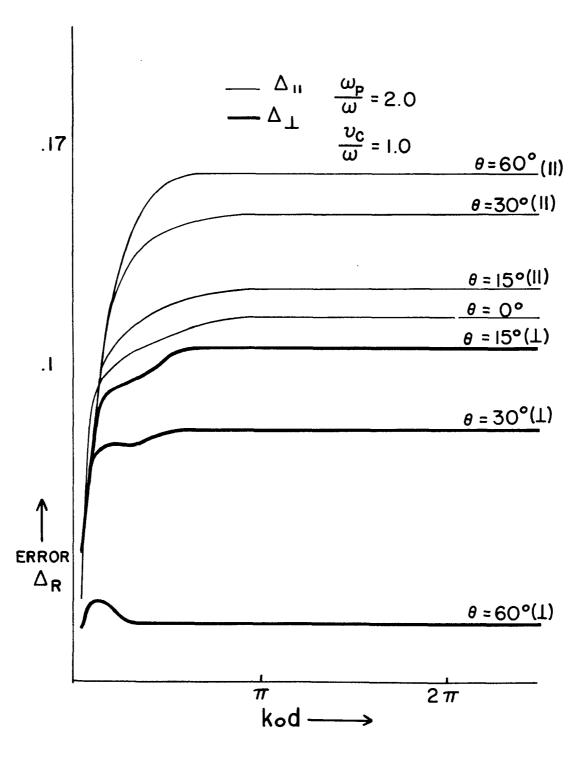


Fig. 8a. First Approximation Reflection Error $(\omega_p/\omega=2)$

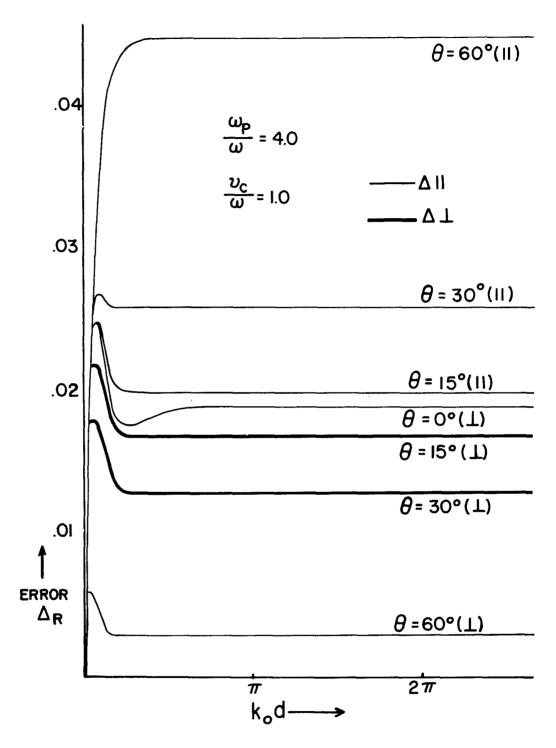


Fig. 8b. First Approximation Reflection Error (ω_p/ω - 4)

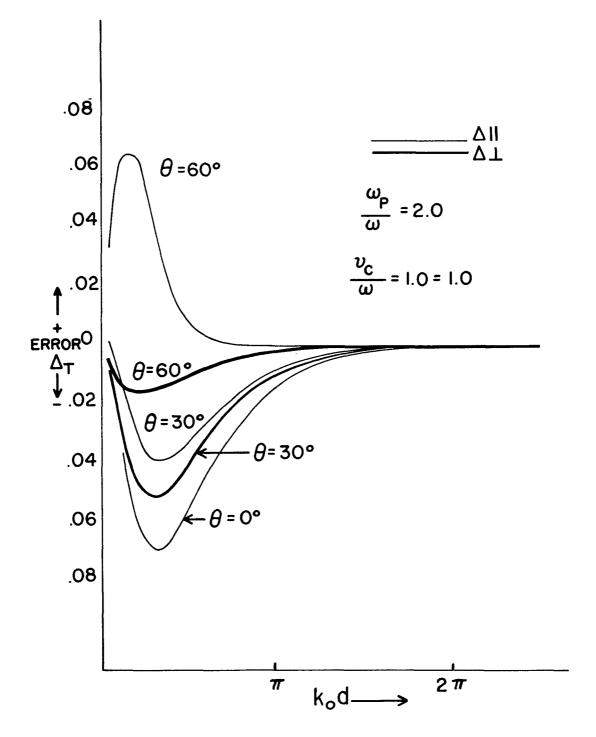


Fig. 9a. First Approximation Transmission Error $(\omega_p/\omega = 2)$

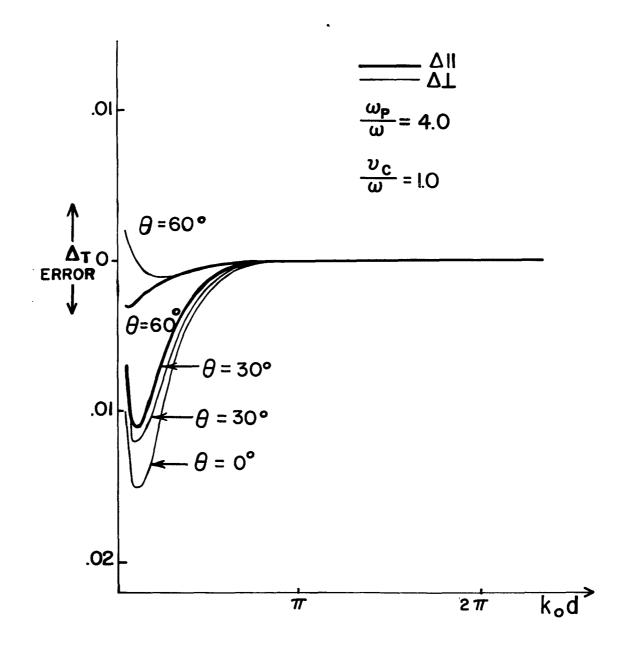
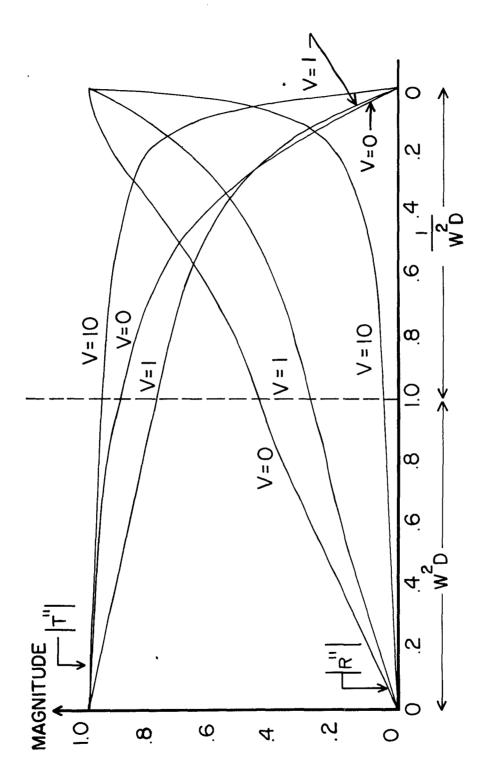


Fig. 9b. First Approximation Transmission Error (ω_p/ω = 4)



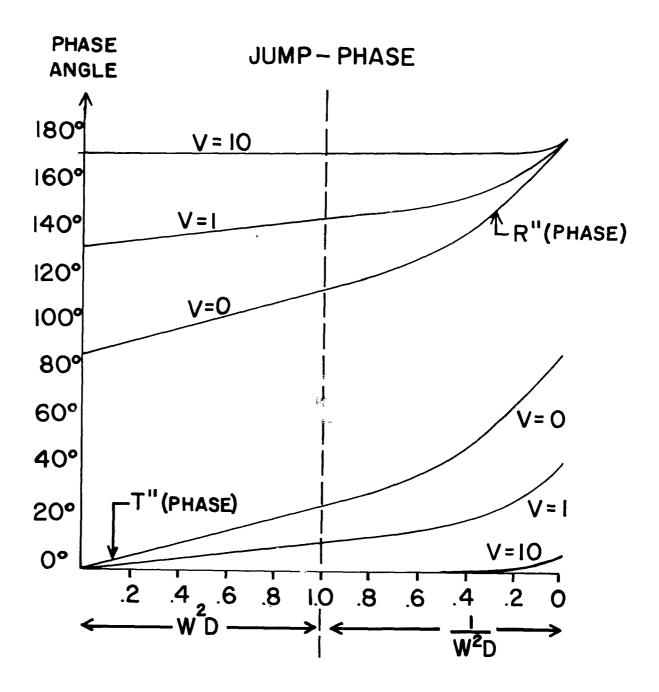


Fig. 10b. Jump Solution (Phase)

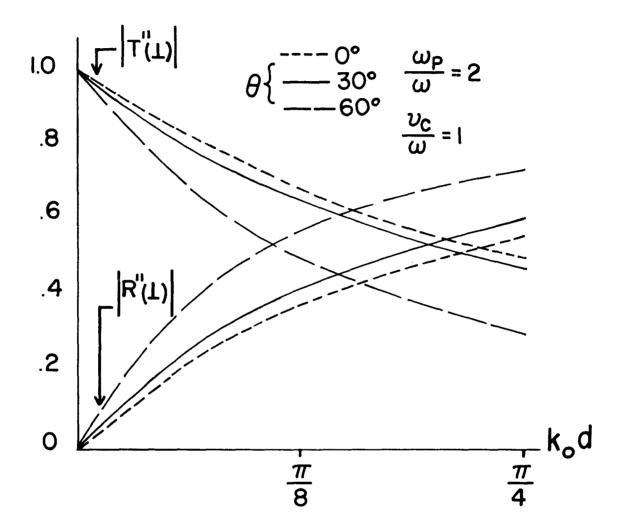


Fig. 11a. Second Approximation, Lossy, $\omega_p/\omega = 2$ (TE Mode)

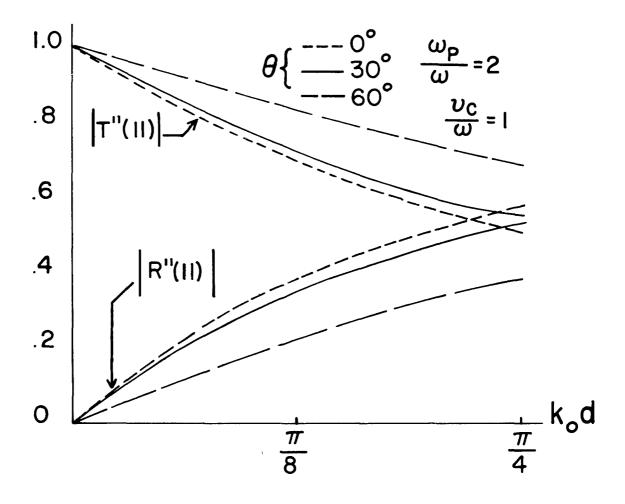


Fig. 11b. Second Approximation, Lossy, $\omega_p/\omega = 2$ (TM Mode)

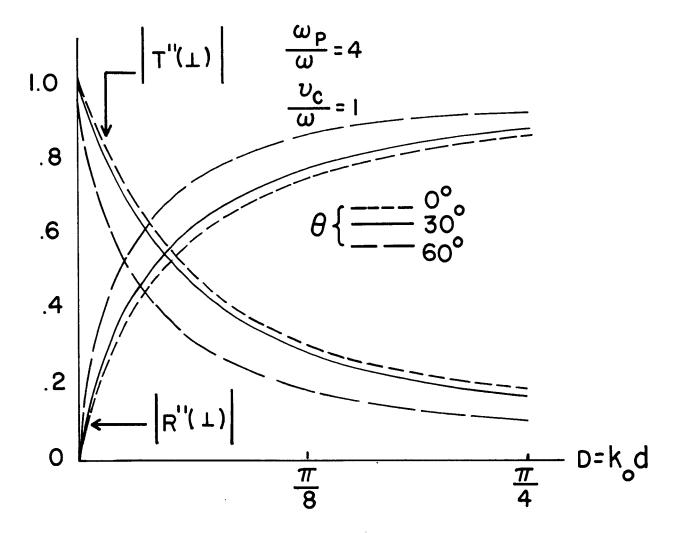


Fig. 12a. Second Approximation, Lossy, $\omega_p/\omega = 4$ (TE Mode)

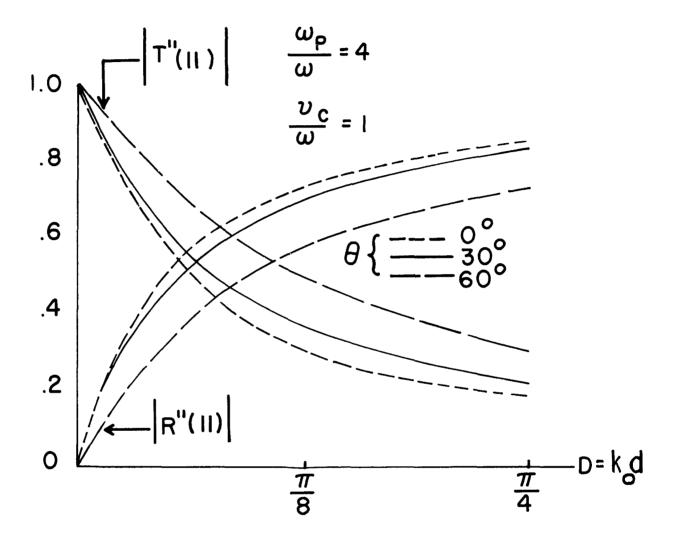


Fig. 12b. Second Approximation, Lossy, $\omega_p/\omega = 4$ (TM Mode)

REFERENCES

- 1. W. Rotman, "Plasma Simulation by Artificial Dielectric and Parallel-Plate Media," I.R.E. Trans. on Antennas and Propagation, Vol. AP10 No. 1, 82-95, (January 1962).
- 2. J. Brown, "Artificial Dielectrics Having Refractive Indices Less than Unity," Proc. I.E.E., Monograph No. 62R, Vol. 100, Part 4, 51-62 (May 1953).
- G. G. McFarlane, "Surface Impedance of an Infinite Parallel-Wire Grid at Oblique Angles of Incidence," <u>Journal I.E.E.</u>, (London), Part IIIA, 93, 1923-1527 (1947).
- 4. J. K. Skwirzynski and J. C. Thackray, "Transmission of Electromagnetic Waves Through Wire Gratings (Theory)," Marconi Review, 22, Second Quarter, 77-90 (1959).
- 5. K. E. Golden and T. M. Smith, "Simulation of a Thin Plasma Sheath by a Plane of Wires," to be published <u>I.E.E.E. Transactions on Nucl. Sci.</u>, Vol. NS11 No. 1 (January 1964).
- 6. E. C. Taylor, "Approximate Boundary Conditions for Plasma Sheaths," to be published, Aerospace Corporation, El Segundo, Calif.
- 7. E. C. Jordan, "Electromagnetic Waves and Radiating Systems," (Prentice-Hall, Inc., New York, N.Y., pp. 143-147 and 155-158, 1950).
- 8. J. R. Wait, "The Electromagnetic Fields of a Dipole in the Presence of a Thin Plasma Sheet," Appl. Sci. Res., 8, Section B, 397-417 (1960).
- 9. R. H. Gold, "Reflection and Transmission of Electromagnetic Waves from Inhomogeneous Magnetoactive Plasma Slabs," TDR-169(3230-11)TN-12 No.(AF 04(695)-169), Aerospace Corp., El Segundo, Calif., p. 22 (March 1963).
- 10. R. G. Jahn, "Interaction of Electromagnetic Waves and Slightly Ionized Gases," Calif. Inst. of Tech., Tech. Note No. 2, p. 15, Pasadena, Calif., Af 49(638)-758, (August 1960).

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